# **Urban Air Mobility**

Advanced Composite Materials Selector Guide





## The Masters of Vertical and Transitional Flight

We chose the dragonfly as our UAM/AAM symbol due to its eVTOL-like exceptional maneuverability and its deep heritage as a symbol of transformation, adaptability, and harmony.

The unique flying characteristics of the dragonfly parallel those of the eVTOL aircraft. By the fast and graceful motion created with their wings, they can lift and land vertically, hover, and fly at varying speeds in all directions.

With more than 5,000 dragonfly species in existence, each varying by their unique wing structures, they illustrate how a diverse range of designs can achieve success. Their power, adaptability, and harmony embody the wholistic approach a Toray partnership offers designers to help create their innovative eVTOL designs.



#### Urban Air Mobility Introduction

#### Your Partner for Tomorrow's Transportation

Toray is ideally positioned to help eVTOL designers and manufacturers create cost-effective prototypes today and prepare for high-rate productions of the future. Our history of successful partnerships in traditional aerospace as well as emerging high-performance industrial and aviation markets provides the confidence needed from a material supplier in an industry with rapid growth and fluid market dynamics. Active collaborations are advancing and demonstrating material and process maturity. A broad portfolio of proven thermoset and thermoplastic materials allow choices and flexibility for applications in this nascent but evolving market. With global locations and an unmatched carbon fiber supply chain, we can ensure your chosen materials are available whenever and wherever they are needed.

#### Why Do eVTOLS Need Composites?

Without advanced composites, eVTOLs would not be possible. Advanced fiber reinforced composite materials are extremely lightweight and are incredibly strong. These inherent material properties enable today's battery technology to effectively manage the power loads needed for vertical and forward propulsion for a variety of vehicle flight ranges.

#### **From Prototype to Production**

Initial designs with low-rate production, minimal nonrecurring cost investment, and a broad knowledge base will drive prototypes toward thermoset solutions.

Our industry-leading thermosets are used on general aviation aircraft, business jets, unmanned areal vehicles (UAV), and traditional vertical lift vehicles. These materials meet the demanding mechanical and environmental requirements and safety standards needed in this early market phase. Initial production rates that mirror current aerospace rates will continue to rely on trusted thermoset composites. As the market matures beyond aerospace production rates, eVTOL manufacturers will have validated and refined designs to demonstrate crashworthiness and impact resistance. Volume production can begin and the transition to higher rates and reduced operating costs will drive a material shift to thermoplastic composites. Processing methods will shift from autoclave and hand lay-up benchmarks to Out of Autoclave (OOA) and Vacuum Bag Only (VBO), ATL/ AFP, stamp and press forming, and continuous compression molding where cut/pick/place steps can be highly automated.

#### **Our Experts Are Here**

Toray's experience with primary and secondary aerospace structures, interior applications, highend automotive, and tooling knowledge means your engineers will be in the best of company. Our Experts Services team will guide your composite material development choices to accommodate the needs of each bespoke eVTOL design.

### **Urban Air Mobility**

**Product Applications** 

#### **Propulsion System**

eVTOL propulsion systems will consist of a combination of rotor blades, propellers, and nacelles structures that must be lightweight and durable. Toray intermediate modulus carbon fiber thermosets are well suited for these applications. Toray Cetex<sup>®</sup> thermoplastics also provide sound dampening characteristics that minimize noise in the surrounding environment as well as in the passenger compartment.



#### Structural

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Structural components such as fuselages, wings, landing gear, and flight control structures such as flaps, ailerons, spoilers/speed brakes, elevators, and ruddervators need high-strength and lightweight materials that can meet both rapid and high-volume manufacturing demands.



#### Panels, Covers, and Communication

Ideal for panels and component covers, Toray thermosets, Cetex® thermoplastics, and bulk molding compounds are lightweight and can be rapidly manufactured in high volume. Toray's industryleading low dielectric products provide low loss and high signal clarity for communication radomes.



#### **Battery System**

Toray's product portfolio of thermosets, Cetex<sup>®</sup> thermoplastics, and bulk molding compounds can be used for battery racks and boxes that must withstand high temperatures while remaining lightweight and incredibly durable.



**Platform Protection** eVTOL exterior surfaces must protect against corrosion and lightning strike. Toray MicroPly\* surface films deliver a strong paintable surface that, when integrated with conductive reinforcements, also provide lightning strike protection. Toray's fiberglass scrim reinforced films offer excellent galvanic barriers.





IGHTNING STRIKE PROTECTION

#### Interiors

Interior eVTOL components must be strong, light, flame-retardant, and visually pleasing while meeting high crashworthiness standards. Toray's Cetex® thermoplastics are well suited for interiors and can be rapidly processed for high-rate production.







SMOOTH SURFACE

								F	orma	ts			Proce	ssing				P	roduc	t Attri	butes				Арр	licatio	ns	
	Resin Matrix	Neat Resin Dry T <sub>g</sub> Onset (DMA)	Cure Temperature and Time	Key Product Characteristics	Out Life # Days	Freezer Life # Months	Design Allowables Database	UDTape	Woven Prepreg	Slit Tape	OOA/VBO	Autoclave	<b>Press Forming</b>	Flexilbe Cure	Freestanding Post cure	AFP/ATL	Toughened	Flame Retardant	Chemical Resistant	Corrosion Resistant	Impact Resistant	High Temperature Peformance	Low Moisture Absorbtion	Structural	Propulsion System	Interiors	Battery System	Panels, Covers, Communication
BT250E-6	Ероху	131°C (268°F)	127°C (260°F)—2 hours	<ul> <li>Aerospace flight qualified</li> <li>71°C (160°F) hot/wet service</li> </ul>	30	12	NIAR NCAMP <sup>1</sup> CHM-17	•	•		•	•	•						•	•				•	•	•		•
2510	Ероху	131°C (294°F)	132°C (270°F)—2 hours	<ul> <li>Qualified to AMS 3960, 3914, and 3915</li> <li>In stock and ready to ship</li> <li>Long freezer life</li> </ul>	28	24	AGATE	•	•		•	•		•					•	•				•	•			•
E732	Ероху	n/a 159°C (318°F) 170°C (338°F)	120°C (248°F)—20 min 140°C (284°F)—10 min 160°C (320°F)—4 min	<ul> <li>Short cure cycles &lt; 20 min</li> <li>Hot-in hot-out press processing</li> </ul>	21	6		•	•		•	•	•	•			•		•	•				•	•	•		•
2511	Ероху	162°C (324°F)	132°C (270°F)—2 hours	<ul> <li>Qualified to AMS 3962</li> <li>Low void content with OOA/VBO</li> <li>Long freezer life</li> </ul>	28	24	CMH-17	•	•	•	•	•		•			•		•	•	•		•	•	•			•
2700	Ероху	163°C (326°F) 200°C (392°F) with post cure	160°C (320°F)—5 minutes (press) 132°C (270°F)—2 hours	<ul> <li>Multi-process flexible system for high volume</li> <li>Short cure cycles &lt; 5 min</li> <li>Low void content and optimized tack</li> </ul>	28	24		•	•	•	•	•	•	•	•	•	•		•	•	•	•	•	•	•			•
TC275-1	Ероху	164°C (327°F) 174°C (345°F) with post cure 183°C (362°F)	135°C (275°F)—6 hours 180°C (356°F)—2 hours post cure (optional) 180°C (356°F)—2 hours	<ul> <li>Aerospace flight qualified</li> <li>Excellent hot/wet strength retention</li> <li>Flexible cure</li> <li>121°C (250°F) hot/wet service</li> </ul>	14	12	NIAR NCAMP <sup>1</sup> CMH-17	•	•	•	•	•	•	•	•	•	•		•	•	•		•	•	•	•	•	•
TC380	Ероху	176°C (349°F) 190°C (374°F) with post cure <sup>3</sup> 201°C (394°F)	135°C (275°F)—6 hours 180°C (356°F)—2 hours post cure (optional) 180°C (356°F)—2 hours	<ul> <li>Aerospace flight qualified</li> <li>Outstanding toughness</li> <li>Excellent balance of OHC and CAI properties</li> <li>121°C (250°F) hot/wet service</li> </ul>	28	12	NIAR NCAMP <sup>2</sup> CMH-17 (in-progress)	•	•	•	•	•	•	•	•	•	•		•	•	•	•	•	•	•	•	•	•
3900	Ероху	204°C (400°F)	177°C (350°F) - 2 hours	<ul> <li>Qualified to AMS 6891</li> <li>In stock and ready to ship</li> <li>Long out life</li> <li>Legacy commercial aerospace material</li> <li>121°C (250°F) hot/wet service</li> <li>Outstanding toughness</li> </ul>	42	24	CMH-17	•	•	•		•				•	•		•	•	•	•	•	•	•			•
3960	Ероху	204°C (400°F)	177°C (350°F) - 2 hours	<ul> <li>Combined with high performance T1100G IM+ fiber</li> <li>Long out life, Extremely long freezer life</li> <li>Excellent balance of CAI and OHC properties</li> <li>121°C (250°F) hot/wet service</li> <li>Outstanding toughness</li> </ul>	42	36	NCAMP (in progress)	•	•	•	•	•	•	•	•	•	•			•	•	•	•	•	•			•

1 - Database is FAA and EASA accepted 2 - Database is FAA accepted 3 - Estimated value

## **Cetex**<sup>®</sup>

								Formats				Processing				Product A			uct Attributes				App	icatio	ns		
	Resin Polymer Temperat Matrix T <sub>m</sub>		Melting Temperature T <sub>m</sub>	Typical Consolidation Temperatures Tp	Key Product Characteristics	Design Allowables Database	UD Tape	Woven Prepreg	Slit Tape	RTL Laminate	Chopped	Weldable/Jointing	Autoclave	Press Forming	AFP/ATL	Toughened	Flame Retardant	<b>Chemical Resistant</b>	<b>Corrosion Resistant</b>	Impact Resistant	High Temperature Performance	Low Moisture Absorption	Structural	Propulsion System	Interiors	Battery System	Panels, Covers, Communication
TC910	PA6	Nylon 6	220°C (428°F)	249–271°C (480–520°F)	<ul> <li>Lower processing temperature option</li> </ul>		•	•	•	•	•	•	•	•	•	•			•	•				•	•		•
TC1000	PEI	Polyetherimide	Amorphous	320–350°C (608–662°F)	<ul> <li>Industry-leading mechanical performance</li> <li>Excellent FST properties</li> </ul>	OEM	•	•	•	•	•	•	•	•	•	•	•	•	•	•		•			•	•	•
TC1100	PPS	Polyphenylene Sulfide	280°C (536°F)	300–330°C (572–626°F)	<ul> <li>Low moisture absorption</li> <li>Ideal for leading edges, beams, clips, and floor panels</li> </ul>	OEM	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•
TC1200	PEEK	Polyetheretherketone	343°C (649°F)	370–400°C (698–752°F)	<ul><li>Continuous processing temperature</li><li>Good high-temperature properties</li></ul>	OEM	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•					
TC1225	LMPAEK	Low-melt Polyaryletherketone	305°C (581°F)	340–385°C (644–725°F)	<ul> <li>Outstanding structural performance</li> <li>Compatible with PEEK for overmolding and welding</li> </ul>	NIAR NCAMP <sup>1</sup> CMH-17	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•
TC1320	PEKK	Polyetherketoneketone	337°C (639°F)	370–400°C (698–752°F)	<ul><li>Outstanding solvent and impact resistance</li><li>Excellent mechanical properties</li></ul>	OEM	•		•		•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•
1 - Database	is FAA accepte	d																									

#### **Toray Cetex® Portfolio**





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## **Bulk Molding Compounds (BMC)**

Thermo	Thermoset BMC										Processing					1	Prod	uct At		Applications							
Resin       Neat Resin         Dry Tg Onset       Cure Temperature and Time         Matrix       (DMA)							Out Life # Days	Freezer Life # Months	Chopped	OOA/VBO	Autoclave	Press Forming	Flexible Cure	Freestanding Post Cure	Toughened	Flame Retardant	Chemical Resistant	Corrosion Resistant	Impact Resistant	High Temperature Performance	Low Moisture Absorption	Lightning Strike Protection	Structural	Propulsion System	Battery System	Panels, Covers, Communication	
MS-4HEpoxy191°C (375°F)138°C (280°F) - 15-30 min 180°C (356°F) - 1-2 hours post cure (freestanding)• Aerospace flight qualified • Epoxy-based structural BMC product • Economical and lightweight aluminum replacement							14	6	•			•		•	•	•	•	•	•		•		•	•	•	•	
Thermo	oplastic	BMC								ßu	Process	ing				Proo	luct Atti	ibutes +	s e				em	Applicat	ions		
	Resin Polymer Temperature Matrix Tm			Typical Consolidation Temperatures Tp	Key Product	Characteri	stics	Chopped	Weldable/Jointi	Autoclave	Press Forming	Toughened	Elomo Dotodou	riame ketardan	Chemical Resist	Corrosion Resist	Impact Resistan	High Temperatu Performance	Low Moisture Absorption	Lightning Strike Protection	Structural	Propulsion Syst	Interiors	Battery System	Panels, Covers, Communication		
MC1100	PPS	Polyphenylen	ne Sulfide	280°C (536°F)	330°C (626°F)	<ul> <li>PPS-based BM0 fibers</li> <li>Fire retardant</li> </ul>	C with SM	and IM	•	•		•	•			•	•	•	•	•	•	•	•	•	•	•	
MC1200	MC1200 PEEK Polyetherketoneketone 343°C (649°F) 385°C (725°F) • PEEK-based fibers • Fire retarda		<ul> <li>PEEK-based BN fibers</li> <li>Fire retardant</li> </ul>	IC with SI	VI and IM	•	•		•	•			•	•	•	•	•	•	•	•	•	•	•				





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								Ρ	rocessi	ng				1	Appl.					
	Resin				OutLife	Freezer I ife	/VBO	clave	s Forming	АТІ	Curable	hened	e Retardant	nical Resistant	osion Resistant	ct Resistant	Temperature ormance	Moisture rption	ining Strike ection	orm Protection
	Matrix	Dry T <sub>g</sub> Onset	Cure Temperature and Time	Key Product Characteristics	# Days	# Months	OOA	Auto	Press	AFP/	Post	Toug	Flam	Chen	Corre	Impa	High Perfo	Low Abso	Light	Platfo
Surfacing Fil	lm Epoxy																			
TC235SF-1	Ероху	119°C (246°F)	121°C (250°F)—60 minutes	<ul> <li>Excellent protective surface finish Available with embedded lightning strike foils</li> <li>Reduces shop floor finishing for productivity savings</li> </ul>	30	12	•	•	•			•		•	•				•	•
Film Adhesiv	ves Epoxy																			
RS-15H	Ероху	99°C (211°F) <sup>1</sup>	93°C (200°F)—6 hours Alternate cures are available	Low temperature curing adhesive	30	12	•	•	•			•		•	•					
TC263	Ероху	110°C (230°F)	121°C (250°F)-2 hours	Ideal for metal or composite bonding	21	12	•	•	•			•		•	•					
TC310	Ероху	157°C (315°F)	177°C (350°F)-2 hours	Ideal for metal or composite bonding	30	12	•	•	•			•		•	•					
Syntactics E	роху																			
EM-3	Ероху	116°C (240°F) <sup>1</sup>	121°C (250°F)—60 minutes	<ul><li>Expanding syntactic coresplice</li><li>High expansion (8–10 x)</li></ul>	14	12		•				•		•	•					
TCF4035	Ероху	140°C (284°F)	130°C (265°F)—2 hours	<ul> <li>Compatible with TAC 121/135°C (250/275°F) curing materials</li> <li>May be post cured for higher T<sub>g</sub></li> </ul>	30	12	•	•			•	•		•	•					
TCF4045	Modified Epoxy	180°C (356°F)	179°C (355°F)—3 hours	Excellent low dielectric constant and loss	14	6	•	•				•		•	•					

1 -  $T_g$  estimated from base resin data



## **Toray AmberTool® Composite Tooling Prepregs**

# **AmberTool**<sup>®</sup>

									Рі	ocessir	g		Product attributes										
	Resin Matrix	Dry Tg Onset (After Postcure)	Min Cure Temp	Typical Cure Time and Temperature	Key Product Characteristics	Out Life # Days	Freezer Life # Months	00A/VBO	utoclave	ress Forming	\FP/ATL	ost Curable	oughened	lame Retardant	chemical Resistant	corrosion Resistant	mpact Resistant	ligh Temperature erformance	ow Moisture \bsorption	ightning Strike rotection			
HX40	Ероху	203°C (397°F)	50°C (122°F)	12 hours at 65°C (149°F)	Large tooling applications	8	12		•	<u> </u>	4	•	-		•	•							
HX42	Ероху	200°C (392°F)	50°C (122°F)	60°C (140°F)—8 hours	<ul><li>Proven system for aerospace</li><li>Excellent surface finish</li></ul>	5	12		•			•			•	•							
HX56	Ероху	185°C (365°F)	40°C (104°F)	50°C (122°F)—8.5 hours	<ul> <li>Improved handleability</li> <li>Excellent surface finish</li> <li>Excellent drape for complex shapes</li> </ul>	60 hours	6		•			•			•	•							



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